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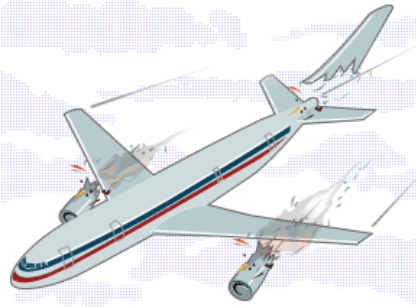
# Delamination Analysis with Cohesive Interface Elements in Finite Element Applications

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## Delamination Risks in Fibre-Reinforced Plastics

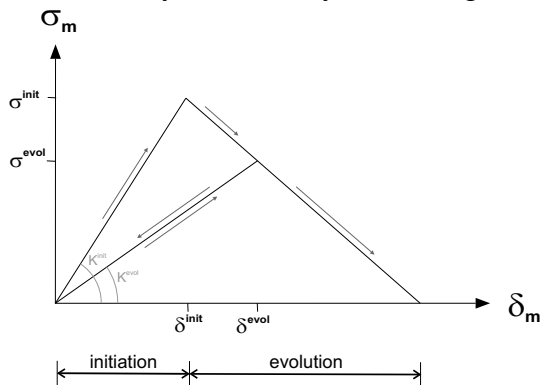
Plastics in combination with reinforcing fibers build low-weight material especially used in aerospace structures. Damage tests and unfortunate aircraft accidents show delamination risks of fibre-reinforced plastics in particular cases of load impact. The crash of Flight 587 in 2001 caused by a debonded tail fin is a famous example.



The finite element method is here an interesting tool to model the delamination process. In this work, the Finite Element Application ABAQUS<sup>®</sup> 6.5-3 was especially used. It provides the therefore applicable cohesive interface elements.

## Interlaminar Damage Modelling

Damage is modelled by degradation of material-stiffnesses. The typical stress-strain relation of non-brittle material is reduced to a basic model with linear elasticity followed by linear degradation.



This bi-linear failure mechanism shows two distinct parts: *initiation* and *evolution*. A scalar damage indicator  $D$  maps the degradation and appears in the actual stress tensor  $\mathbf{t}$  in order to relate to the initial stress tensor  $\bar{\mathbf{t}}$ .

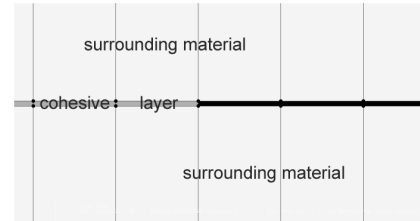
$$t_n = \begin{cases} (1 - D)\bar{t}_n & , \bar{t}_n \geq 0 \\ \bar{t}_n & , \text{other} \end{cases}$$

$$t_s = (1 - D)\bar{t}_s$$

$$t_t = (1 - D)\bar{t}_t$$

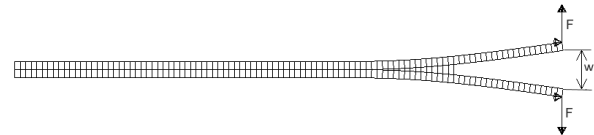
## How to include Cohesive Elements

Cohesive elements can be inserted where cracks are expected. They build an interfacial zone between two parts which are supposed to separate.

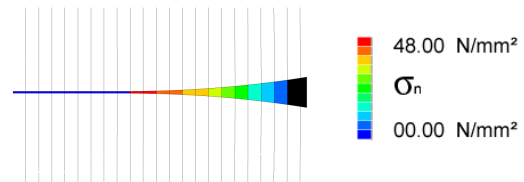


## Application to a Test Specimen

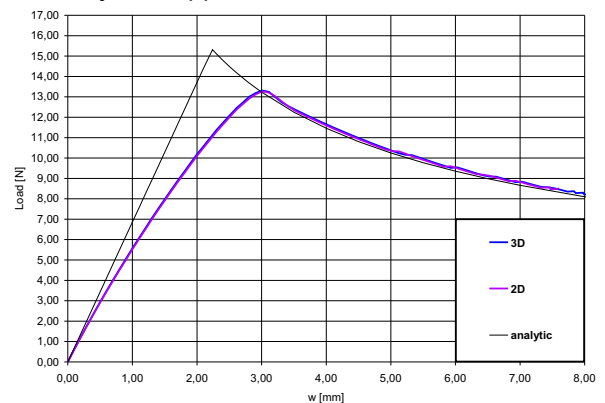
A numerical simulation of a delamination test specimen called Double Cantilever Beam was executed in ABAQUS<sup>®</sup>.



A view on the stresses near the crack tip shows degradation of the cohesive layer by increment of  $F$ .



During the simulations convergence problems occurred that only were overcome by assuming viscous material behaviour to the cohesive elements. The achieved load/crack-opening curve  $F(w)$  shows good agreement to an analytical approach.



Further development and improvement of cohesive elements are necessary for an application to more complex problems of practical engineering.